Formalizing Hybrid Systems with Event-B

Jean-Raymond Abrial, Wen Su, Huibiao Zhu

August 2012

Prologue

Using B Formal Method in Industry

- Fully automatic train systems:
 - Paris metro line 14 (October 1998)

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- Roissy airport shuttle (March 2007)
- More train applications

Line length	8.5 km
Number of Stops	8
Time interval between two trains	115 s
Speed	40 km/h
Number of trains	17
Passengers per day	350,000

Line length	3.3 km
Number of Stops	5
Time interval between two trains	105 s
Speed	26 km/h
Number of trains	14
Passengers per hour	2,000

	Paris	Roissy
Number of final ADA lines (from B)	86,000	158,000
Number of proofs	27,800	43,610
Percentage of interactive proofs	8.1	3.3
Interactive proofs in Man.Month	7.1	4.6

- Man.month calculated with:
 - 15 interactive proofs per man.day
 - 21 days in a month
- In both cases, no unit tests and no integration tests
- Reinforcing global tests (catastrophic scenarios)

-Important differences in the software requirements:

- Paris: specially done for the project
- Paris: adaptation from O'Hare Airport (problems)

City	Line	Service	Driverless
Algiers	1	2011	No
Barcelona	9	2007	Yes
Budapest	4	2013	Yes
Caracas	4	2004	No
Helsinki	1	2013	No
Hong Kong	ТКО	2001	No

Similar Applications by Siemens (2)

City	Line	Service	Driverless
Mexico	В	2000	No
New York	Canarsie	2006 No	
	PATH	2014	No
Paris	14	1998	Yes
	3	2009	No
	1	2011	Yes
	5	2012	No

City	Line	Service	Driverless
Rennes	В	2018	Yes
Roissy CDG	1 2	2007 2007	Yes Yes
San Juan	2	2004	No
Sao Paulo	ТКО	2001	Yes

Contact: <Jean-Marc.Meynadier@siemens.com>

System	City	Service	Size	Language	Driverless
KVB	French Trains	1993	30000	ADA	No
CDTC	Cairo	1996	3000	Modula2	No
SACEM	Paris (RER B)	1996	2500	Modula2	No
ACSES	AMTRACK (USA)	2002	14500	ADA	No

System	City	Service	Size	Language	Driverless
Urbalis 200	Shanghai	2003	30000	ADA	No
	New Dehli				
	Seoul				
	Daegu				
	Incheoun				
	Madrid				
	Santiago				
	Cairo	2013			
	Bangalore				
	Calcutta				

System	City	Service	Size	Language	Driverless
Urbalis 400	Shanghai	2008	100000	ADA	No
	Beijing				Yes
	Chenzen				No
	Sao Paulo	2013			Yes
	Mexico				No
	Milano				No
	Toronto				No
	Wuhan				No

Contact: <Luis-Fernando.Mejia@transport.alstom.com>

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- Hence the notion of hybrid systems

where both discrete and continuous transitions can occur

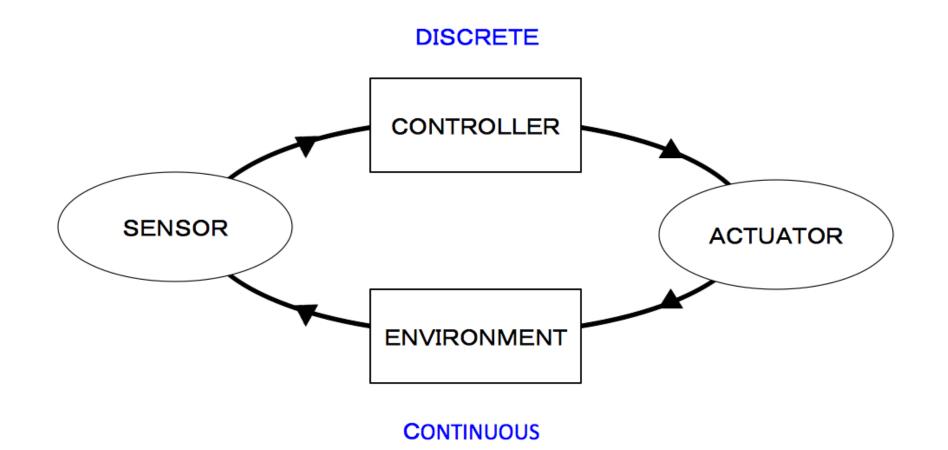
- A piece of software, the controller, manages an environment

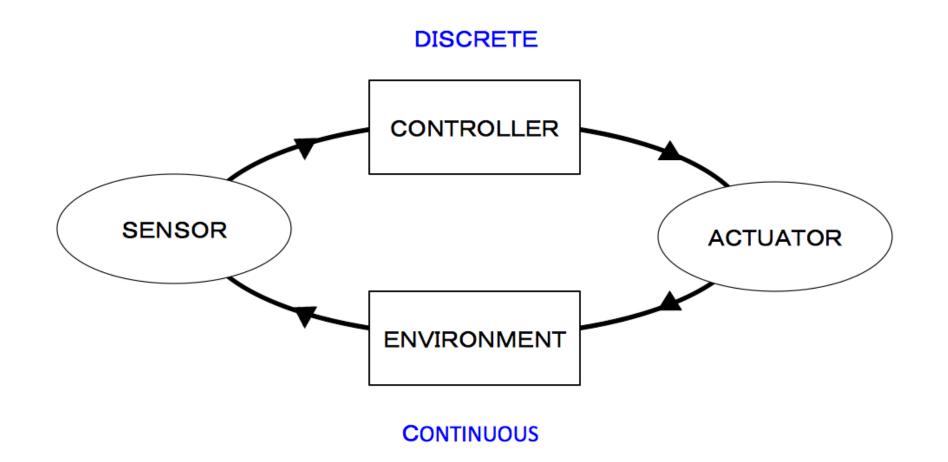
- A piece of software, the controller, manages an environment

- Controller is linked to environment by sensors and actuators

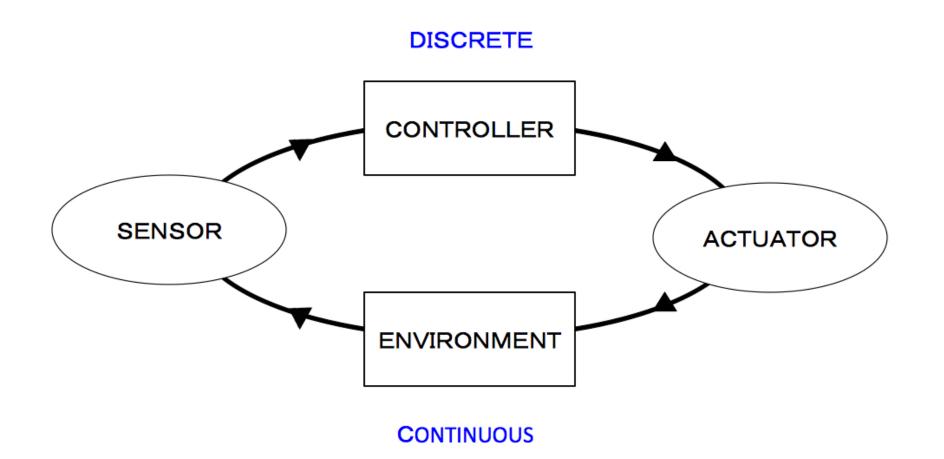
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- Controller works from time to time in a **DISCRETE** fashion

- A piece of software, the controller, manages an environment
- Controller is linked to environment by sensors and actuators
- Controller works from time to time in a **DISCRETE** fashion
- While environment evolves in a CONTINUOUS way.





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- We want to develop models of such closed systems
- We have thus to cope with both discrete and continuous evolutions

Example

- Continuous physical environment:
 - a train defined by its position, speed, and acceleration

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- Discrete controller:

a driver changing from time to time the acceleration of the train

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- Discrete controller:

a driver changing from time to time the acceleration of the train



- Goal: to control the speed of the train (station or another train)

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Distributed Cooperation with Action Systems

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R.J. Back, L. Petre, and I. Porres.

Generalizing Action Systems to Hybrid Systems.

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R.J. Back, C. Cerschi Seceleanu, and J. Westerholm.

Symbolic Simulation of Hybrid Systems.

APSEC'02, 2002.

Formalizing Hybrid Systems with Event-B

ABZ Conference. Pisa, June 2012

Complementary Methodologies for Developing Hybrid Systems with Event-B

Accepted at ICFEM 2012. Kyoto, November 2012

- Discrete variables together with continuous variables

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- Continuous variables are time functions as in Action System
- We are interested in the immediate future of continuous variables
- Discrete systems as an abstraction of continuous ones
- We thus use refinement from a discrete to a continuous system

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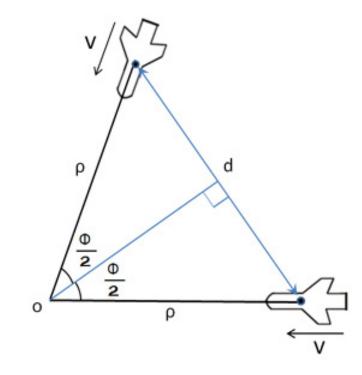
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- Description:
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- Examples developed and fully proved with the Rodin Platform
- These examples show complete analytical solutions

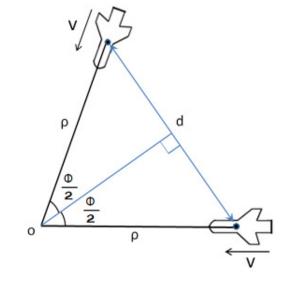
Example 1

- Two aircrafts are flying at the same altitude and speed

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- They might converge (collision) at some point o

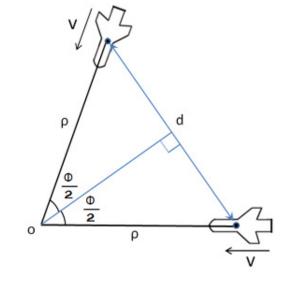
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- The distance between aircrafts is as follows:

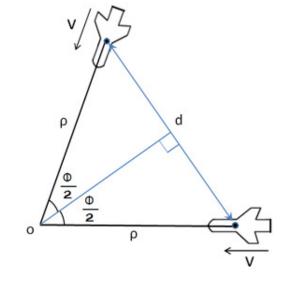
$$d=2
ho\sinrac{\phi}{2}$$



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- Their distance must always be greater than or equal to a constant p

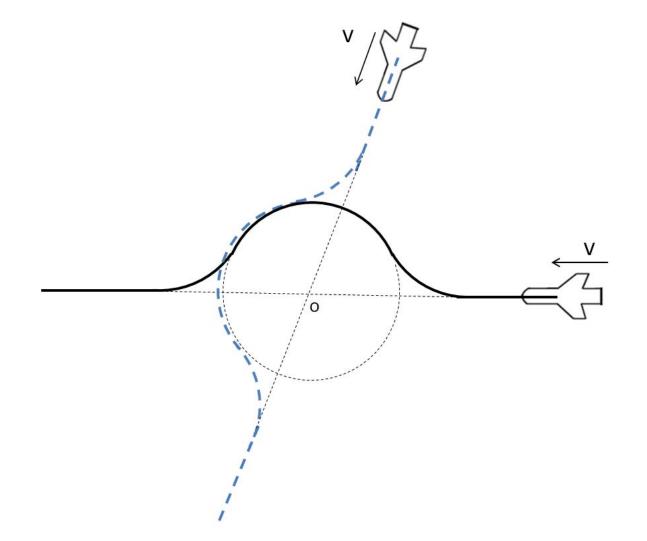


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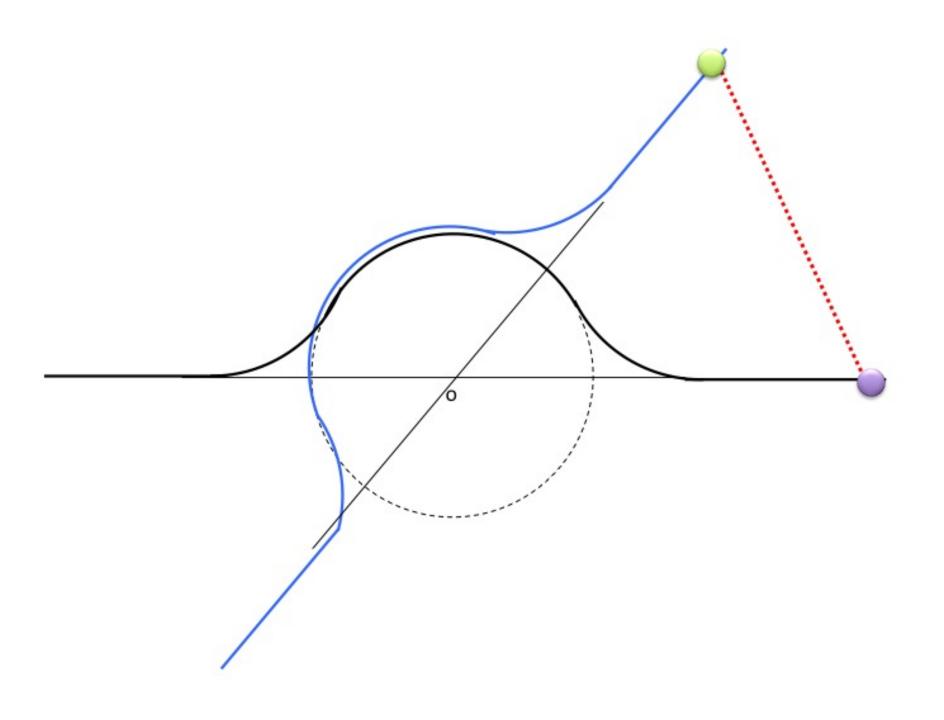
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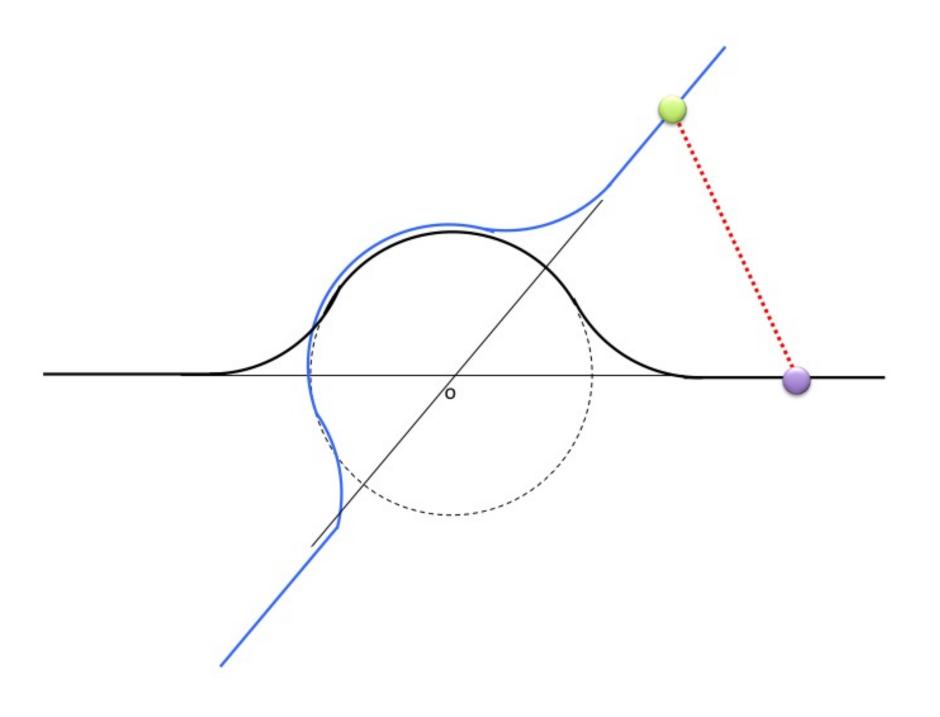
- Their distance must always be greater than or equal to a constant p
- Goal: we want to find a solution to avoid the collision

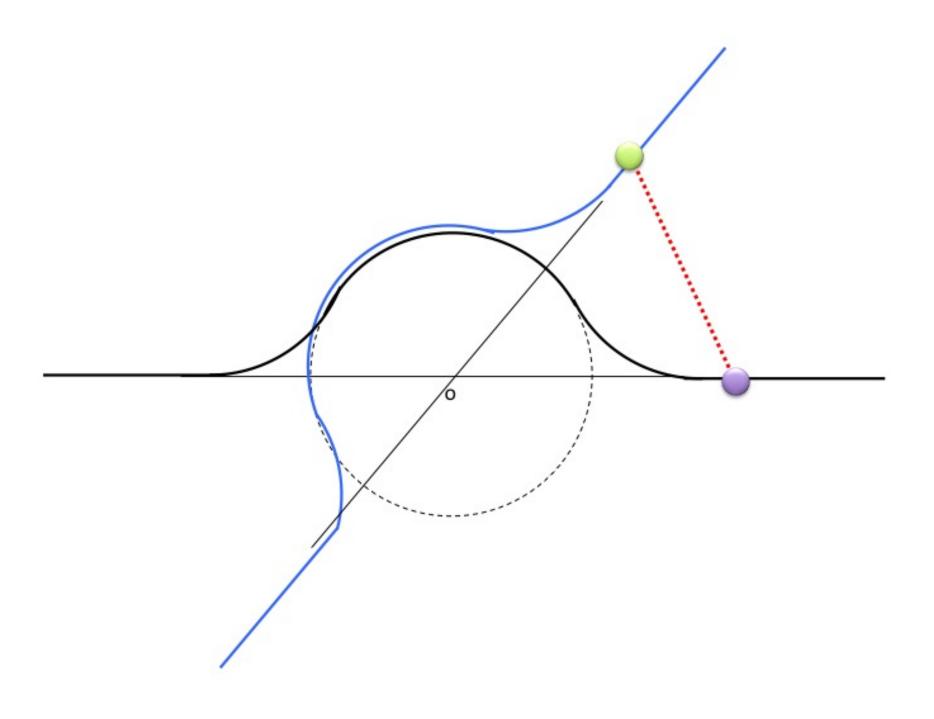
The Solution

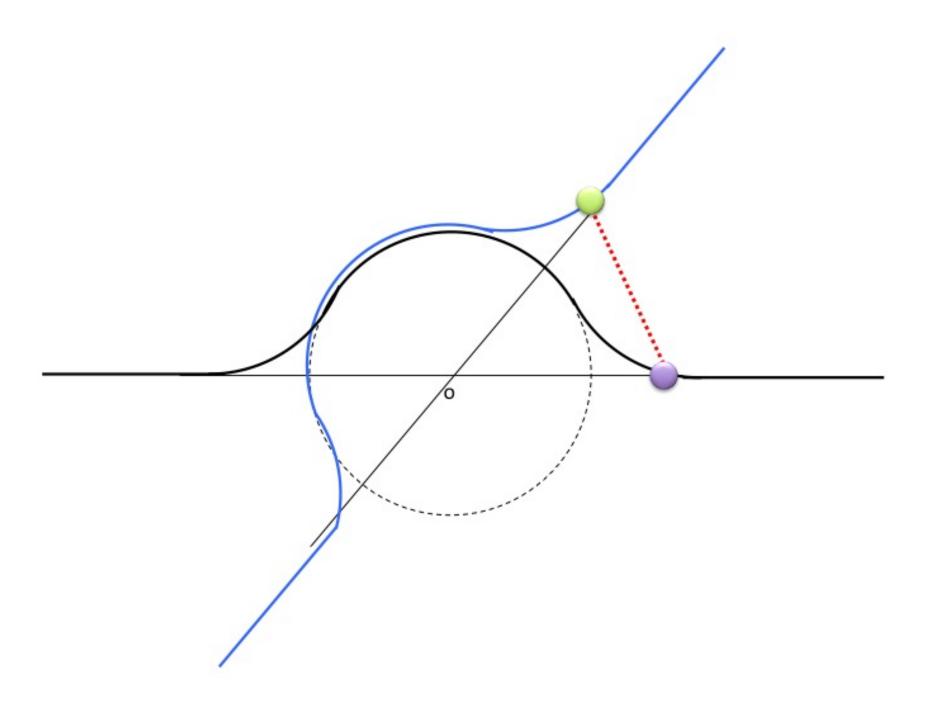


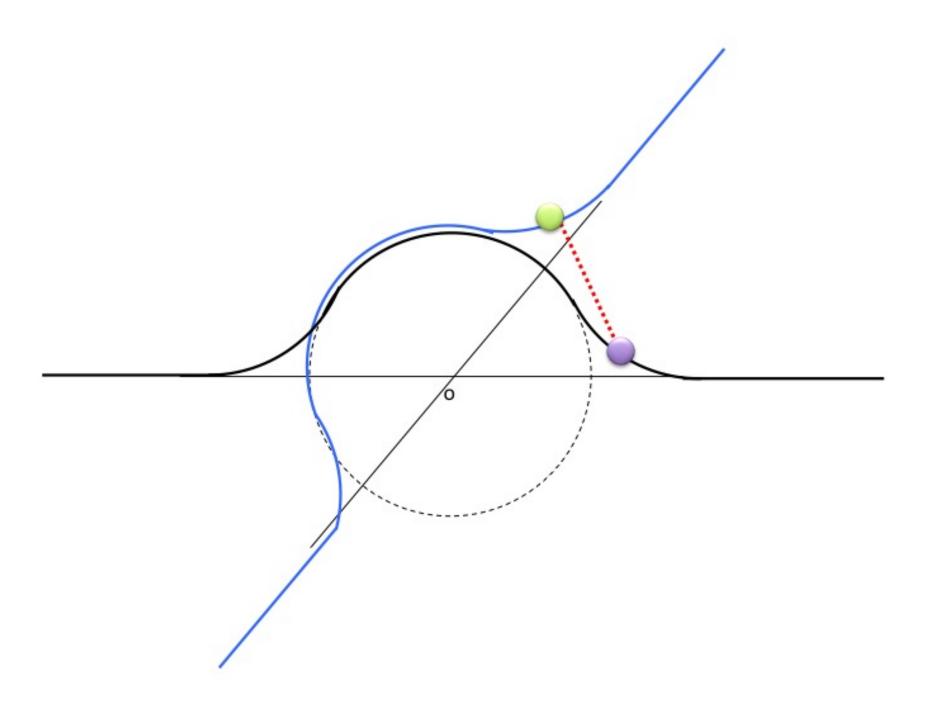
- The radius *r* of this circle will be determined later
- Both aircrafts continue to fly at the same speed during the maneuver

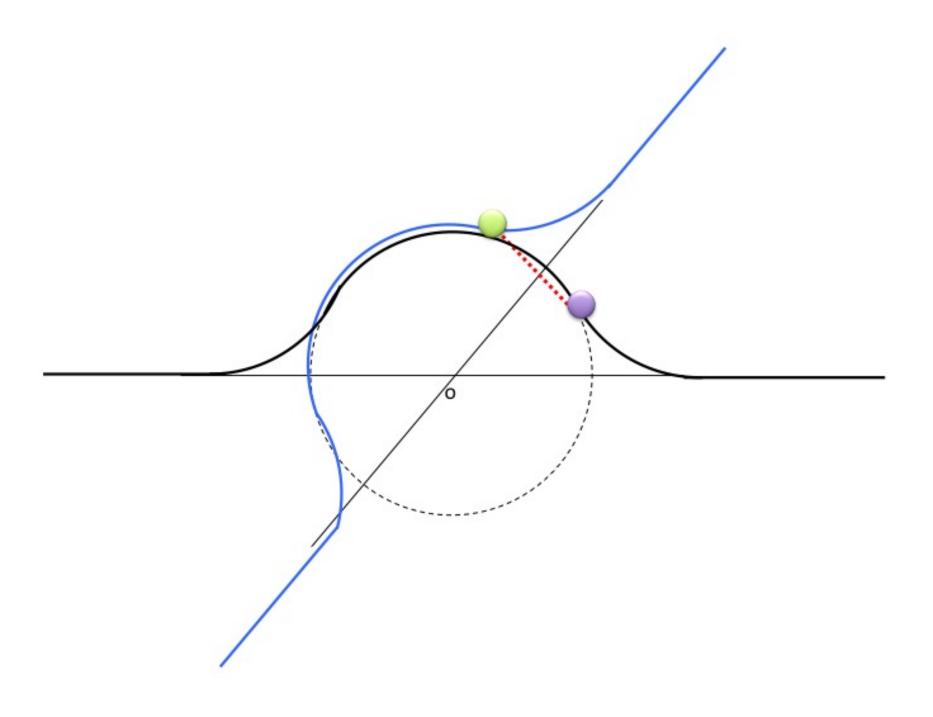


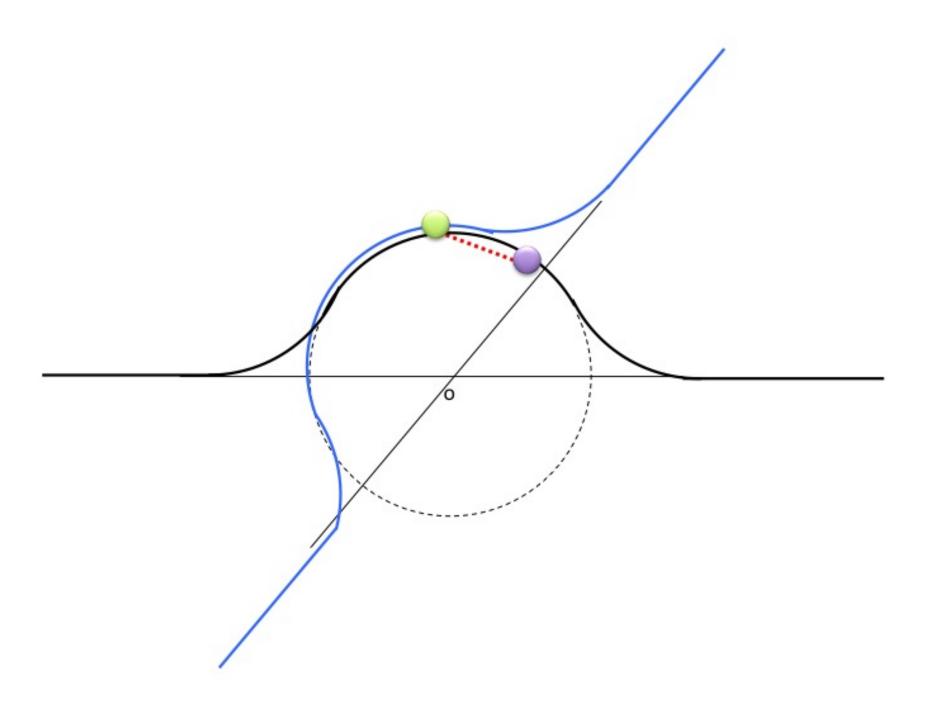


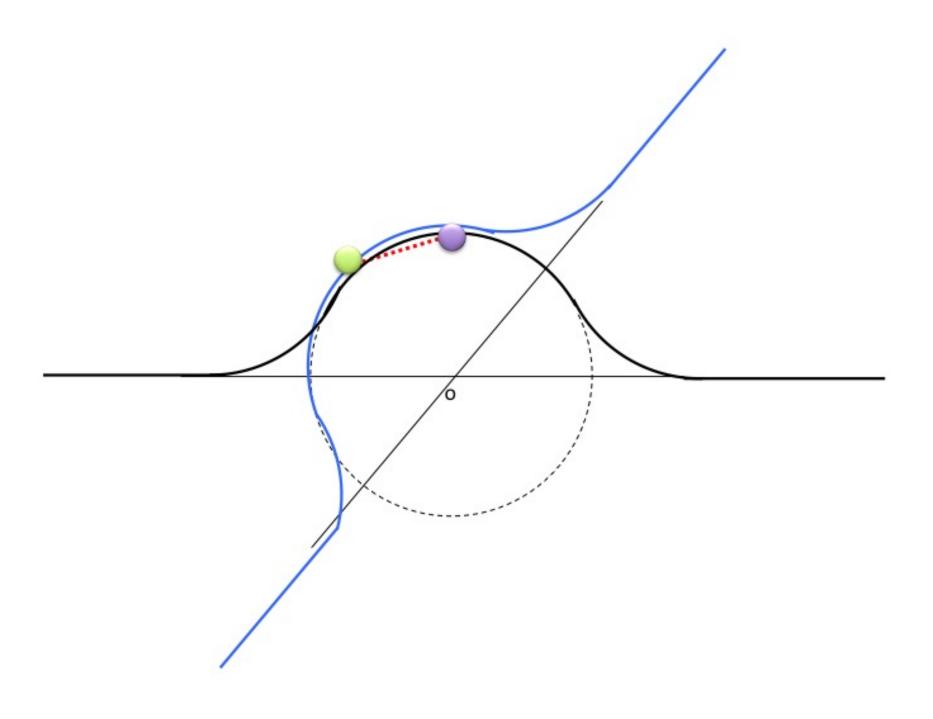


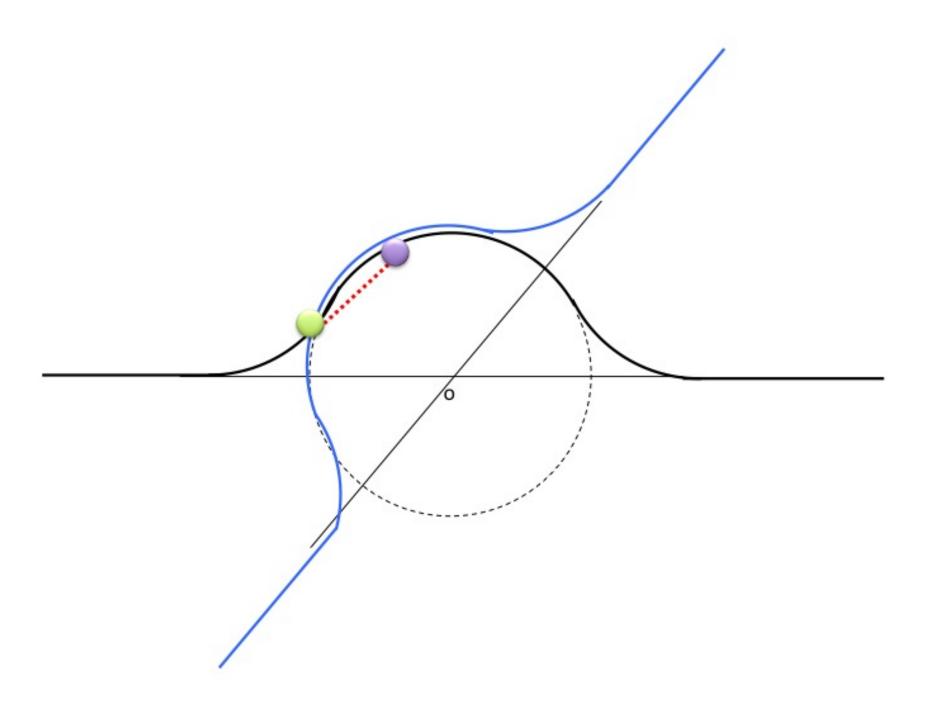


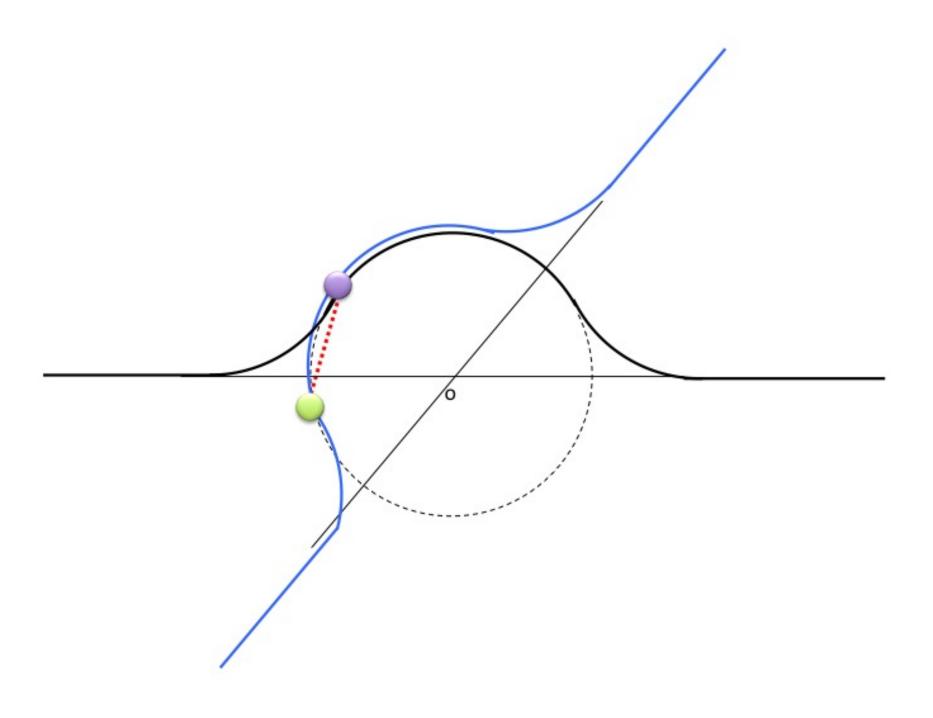


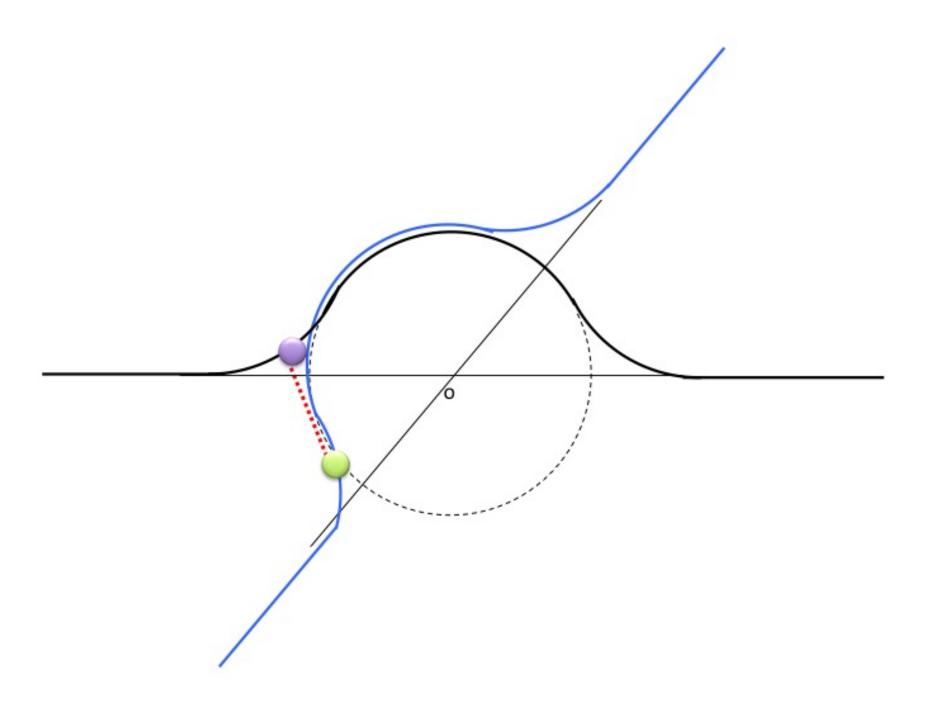


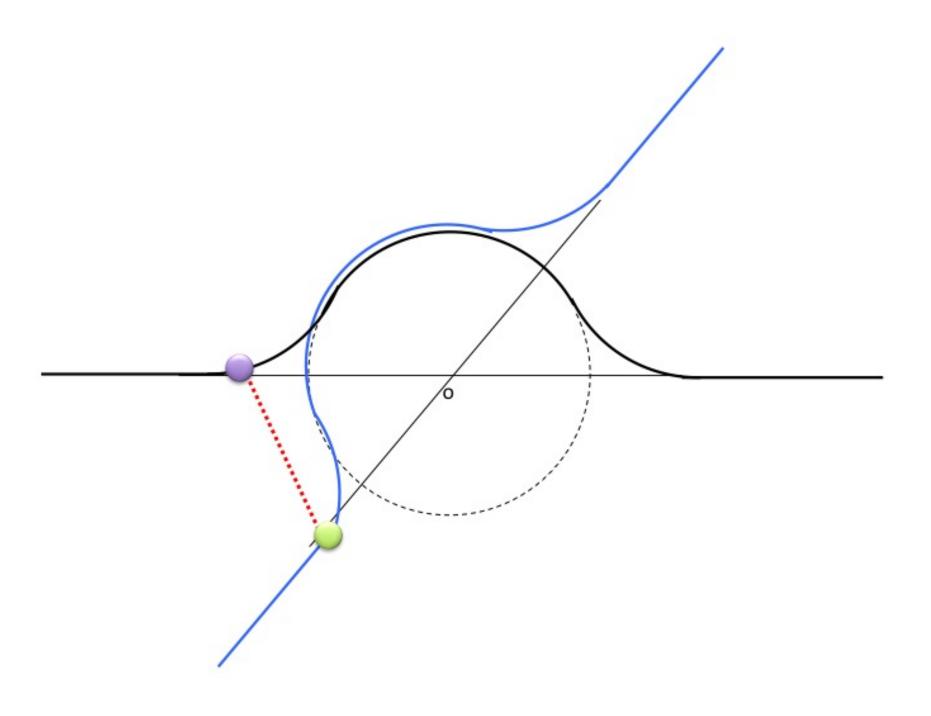


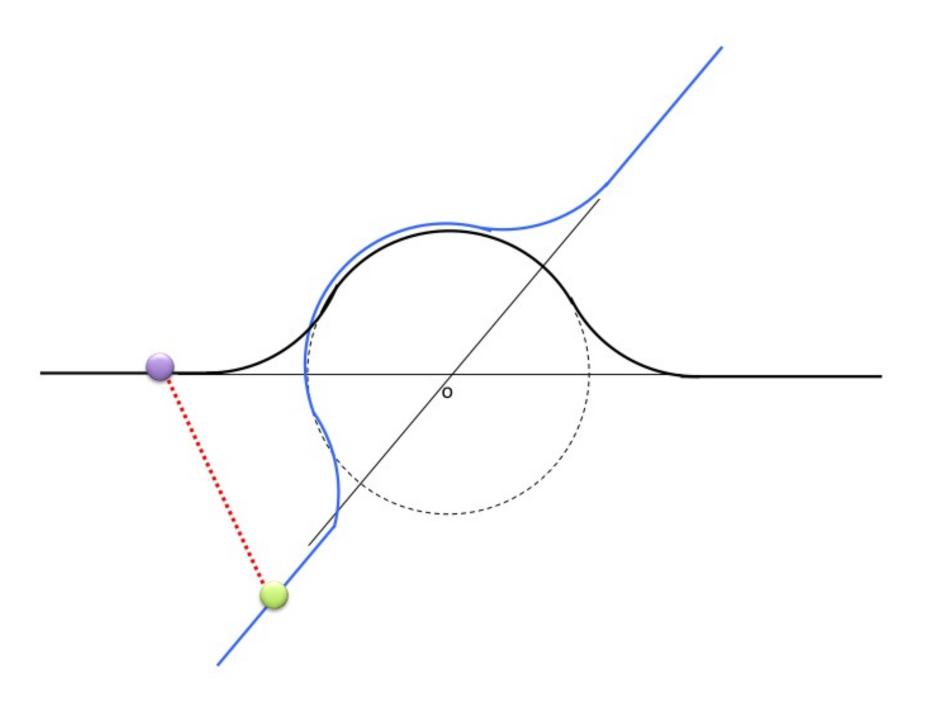


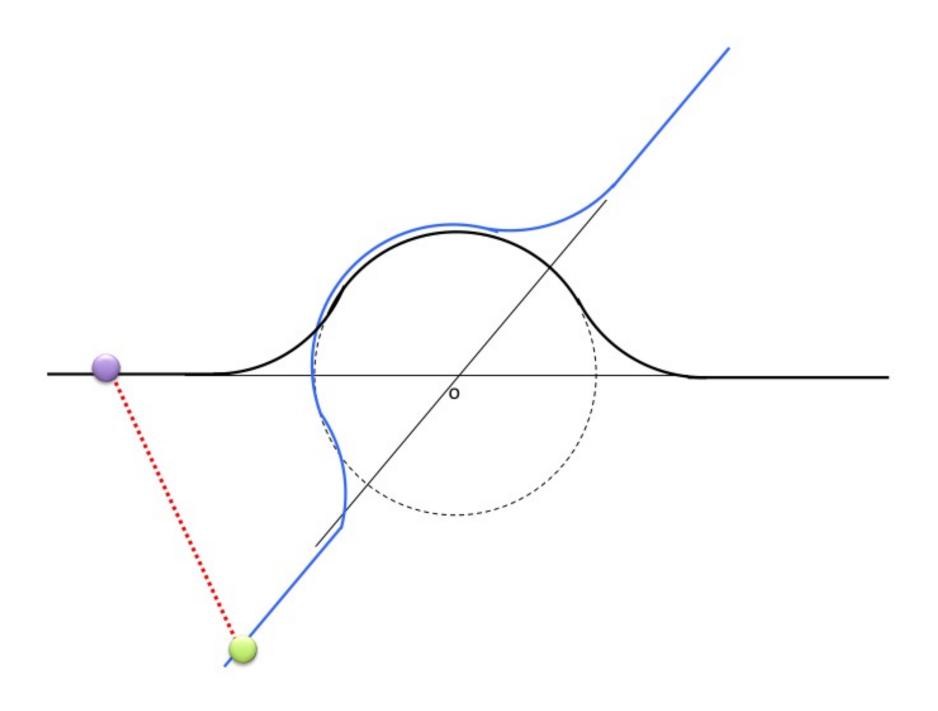












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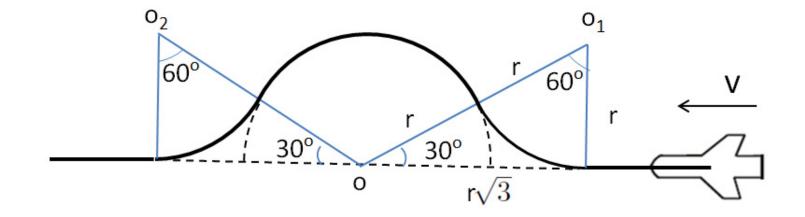
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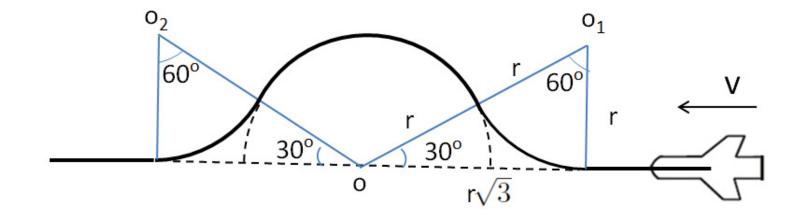
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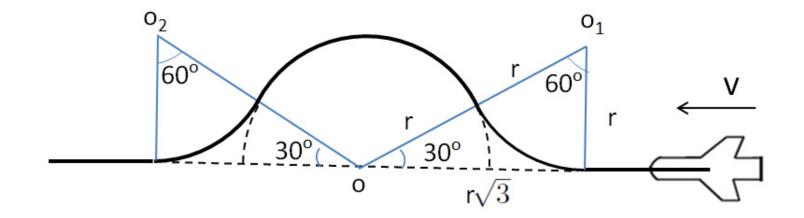
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- We must have then:
$$\frac{p}{2\sinrac{\phi}{2}} \leq r$$

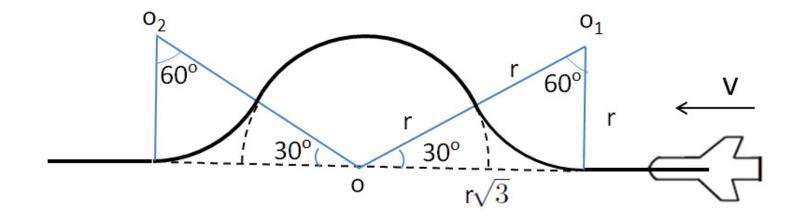




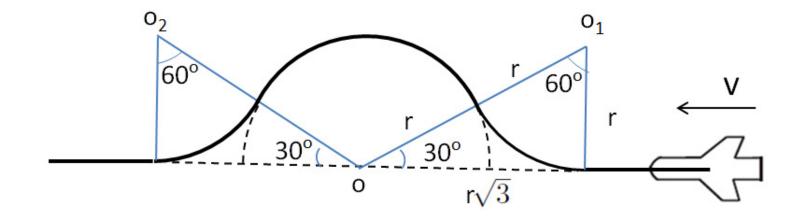
- Both aircrafts fly as indicated on this figure



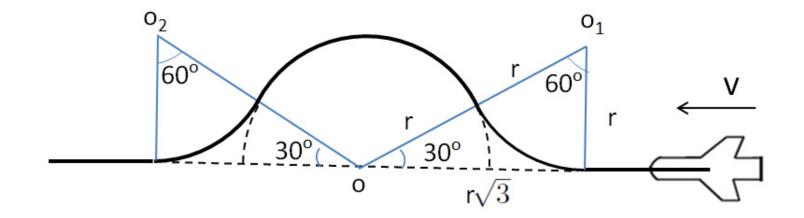
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- They start the maneuver when at a distance $r\sqrt{3}$ from the point o



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 ho_i}{\sqrt{3}}$



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- We have then:

$$rac{p}{2\sinrac{\phi}{2}} \leq r \leq rac{
ho_i}{\sqrt{3}}$$

- Here is again the possible interval for the radius r of the circle:

$$rac{p}{2\sinrac{\phi}{2}} \leq r \leq rac{
ho_i}{\sqrt{3}}$$

- We must have then the following for the constants ρ_i , ϕ , and p:

$$2
ho_i \sin rac{\phi}{2} \geq p\sqrt{3}$$

- ϕ is the angle of the two trajectories
- ρ_i is the initial distance of the two aircrafts to the collision point o
- p is the minimal safety distance between the two aircrafts

```
axm1: \phi \in 0 \dots \piaxm2: \rho_i \in \mathbb{R}^+axm2: p \in \mathbb{R}^+axm4: 2\rho_i \sin \frac{\phi}{2} \ge p\sqrt{3}
```

- In this initial model, we are still discrete

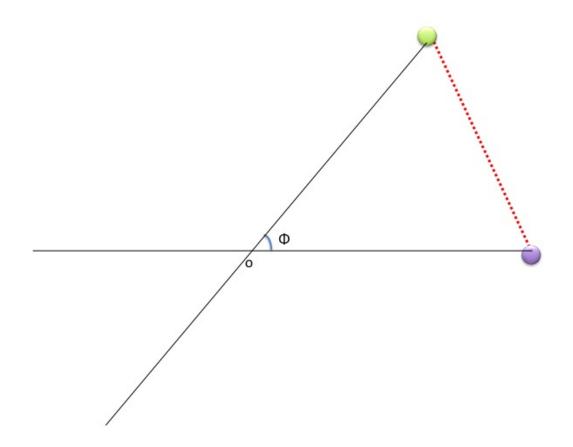
- *phase* corresponds to the various discrete events
- ho is the common distance of the aircrafts to the collision point o
- *r* is the circle radius

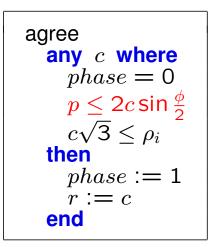
inv1: $phase \in \{0, 1, 2, 3, 4, 5\}$ inv2: $\rho \in \mathbb{R}^+$ inv3: $r \in \mathbb{R}^+$ inv4: $2\rho \sin \frac{\phi}{2} \ge p$

- **inv4** is the safety invariant: the minimal authorized distance is p

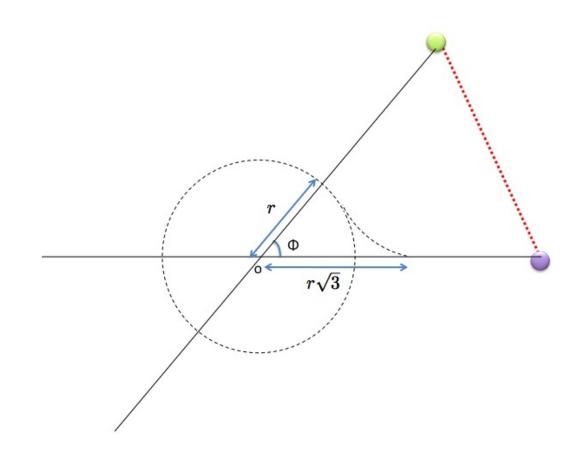
- INIT: initialisation
- agree: choose the radius of the circle
- start: start the maneuver
- enter: entering the circle
- cycle: move on the circle
- leave: leaving the circle

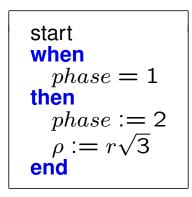
$$\begin{array}{c} \text{INIT} \\ \textbf{begin} \\ \rho := \rho_i \\ phase := 0 \\ r :\in \mathbb{R}^+ \\ \textbf{end} \end{array}$$



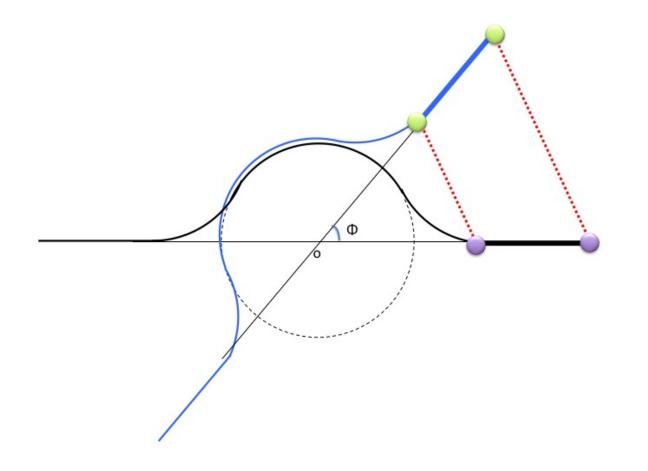


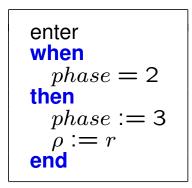
Choosing the radius r of the circle



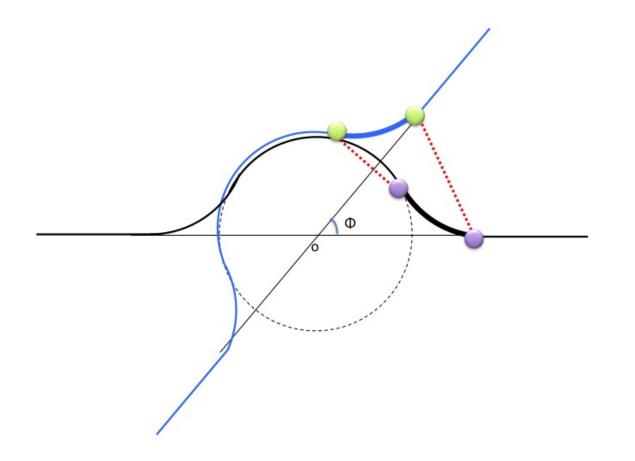


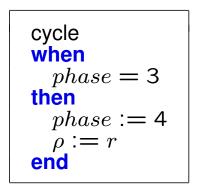
 ρ goes from ρ_i to $r\sqrt{3}$



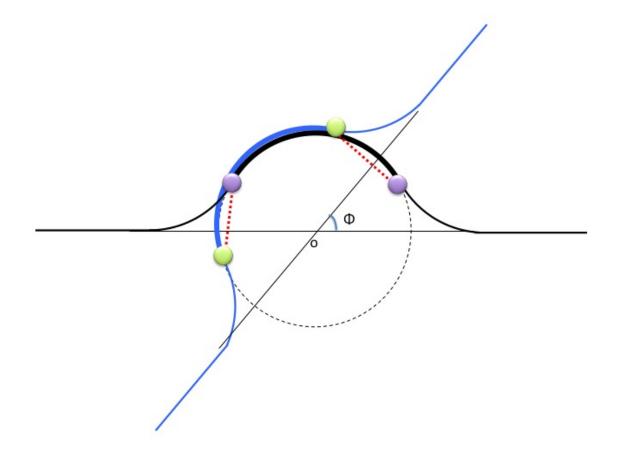


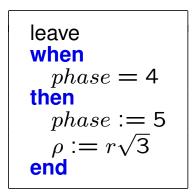
 ρ goes from $r\sqrt{3}$ to r



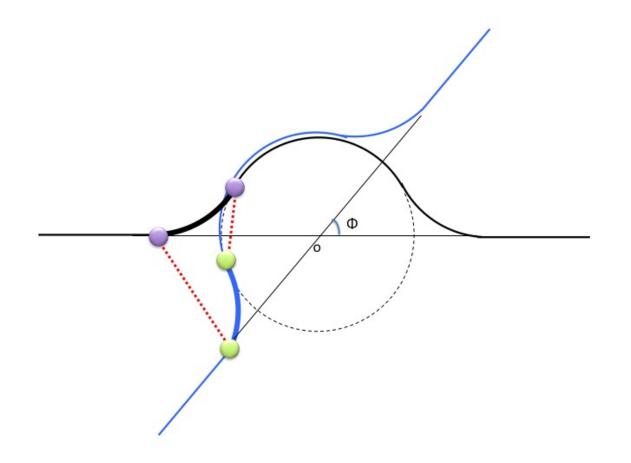


 ρ goes from r to r





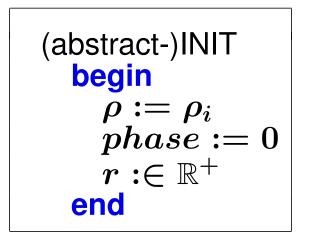
 ρ goes from r to $r\sqrt{3}$



- We introduce the intermediate continuous parts
- We replace ρ by $\rho_{-}c$ (that is $\rho_{-}continuous$)
- We introduce *now*, the present time

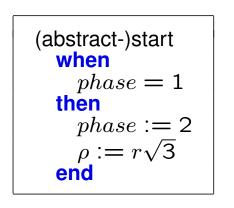
inv1_1: $\rho_c \in \mathbb{R}^+ \oplus \mathbb{R}$ inv1_2: $now \in dom(\rho_c)$ inv1_3: $\rho = \rho_c(now)$ inv1_4: $\forall t \cdot t \in dom(\rho_c) \Rightarrow 2\rho_c(t) \sin \frac{\phi}{2} \ge p$

- **inv1_3** is the gluing invariant
- **inv1_4** generalises the previous invariant: $2\rho \sin \frac{\phi}{2} \ge p$



$$\begin{array}{l} (\text{concrete-})\text{INIT} \\ \textbf{begin} \\ \rho_c:=\{0\mapsto\rho_i\} \\ phase:=0 \\ r:\in\mathbb{R}^+ \\ now:=0 \\ \textbf{end} \end{array}$$

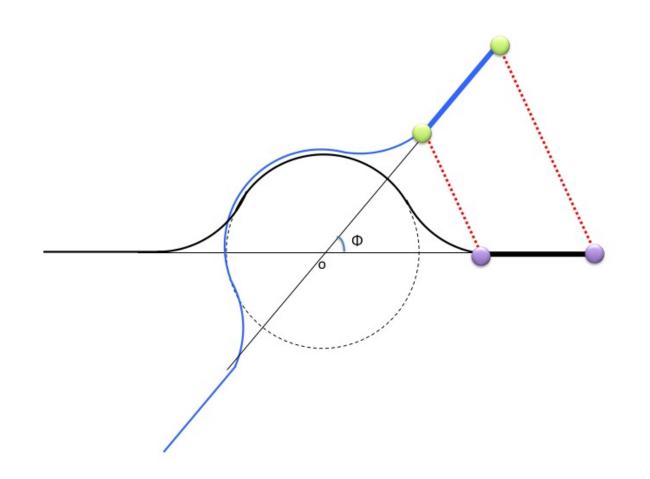
agree
any
$$c$$
 where
 $phase = 0$
 $p \leq 2c \sin \frac{\phi}{2}$
 $c\sqrt{3} \leq \rho_i$
then
 $phase := 1$
 $r := c$
end



(concrete-)start
when

$$phase = 1$$

then
 $phase := 2$
 $\rho_{-}c := \lambda t \cdot t \in now .. now + \frac{(\rho_i - r\sqrt{3})}{v} | \rho_i - v(t - now)$
 $now := now + \frac{(\rho_i - r\sqrt{3})}{v}$
end



start
when

$$phase = 1$$

then
 $phase := 2$
 $\rho_{-}c := \lambda t \cdot t \in now \dots now + \frac{(\rho_i - r\sqrt{3})}{v} | \rho_i - v(t - now)$
 $now := now + \frac{(\rho_i - r\sqrt{3})}{v}$
end

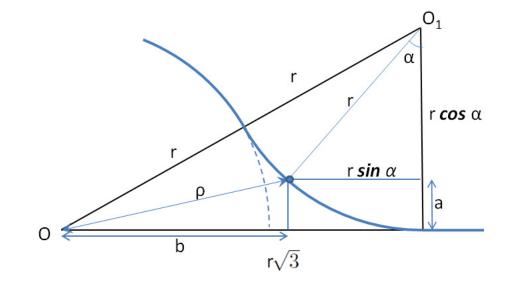
- $ho_{-}c(now)=
ho_{i}$

-
$$ho_{-}c(now+rac{(
ho_i-r\sqrt{3})}{v})=r\sqrt{3}$$

- $ho_{-}c$ decreases linearly from ho_{i} to $r\sqrt{3}$

- $\frac{(
ho_i - r\sqrt{3})}{v}$ is the time it takes to fly from ho_i to $r\sqrt{3}$

Computing ρ **During First Part of Maneuver**



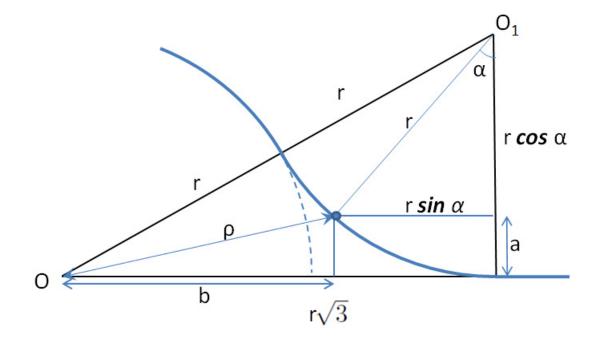
$$\rho^{2} = a^{2} + b^{2}$$

$$= r^{2}(1 - \cos \alpha)^{2} + r^{2}(\sqrt{3} - \sin \alpha)^{2}$$

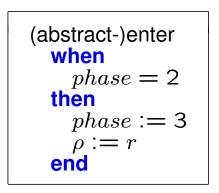
$$= r^{2}(5 - 4\cos(\frac{\pi}{3} - \alpha))$$

$$\rho = r\sqrt{5 - 4\cos(\frac{\pi}{3} - \alpha)}$$

- ρ decreases from $r\sqrt{3}$ to r when α goes from 0 to $\frac{\pi}{3}$.



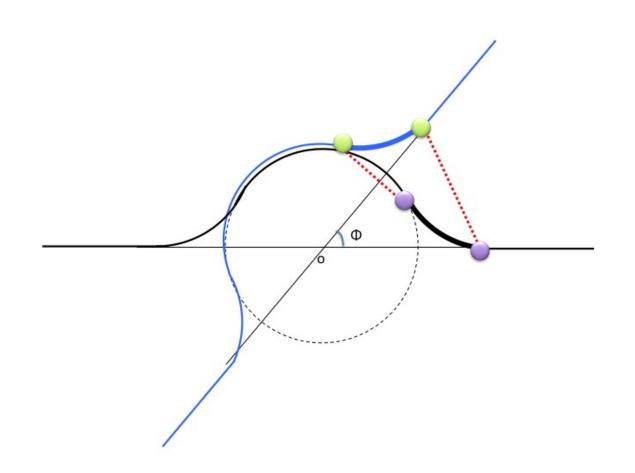
- The angle α increases from 0 to $\frac{\pi}{3}$ during this phase
- The distance is $\frac{\pi r}{3}$
- The time to cover this distance is thus $\frac{\pi r}{3v}$
- We have: $lpha = rac{v(t-now)}{r}$



concrete-)enter
when

$$phase = 2$$

then
 $phase := 3$
 $\rho_{-c} := \lambda t \cdot t \in now ... now + \frac{\pi r}{3v} | r\sqrt{5 - 4\cos(\frac{\pi}{3} - \frac{v(t-now)}{r})}$
 $now := now + \frac{\pi r}{3v}$
end



enter
when

$$phase = 2$$

then
 $phase := 3$
 $\rho_{-}c := \lambda t \cdot t \in now \dots now + \frac{\pi r}{3v} | r\sqrt{5 - 4\cos(\frac{\pi}{3} - \frac{v(t-now)}{r})}$
 $now := now + \frac{\pi r}{3v}$
end

$$-
ho_{-}c(now) = r\sqrt{5-4\cos{rac{\pi}{3}}} = r\sqrt{5-rac{4}{2}} = r\sqrt{3}$$

- $ho_{-}c(now + rac{\pi r}{3v}) = r\sqrt{5 4\cos(rac{\pi}{3} rac{v\pi r}{r3v})} = r\sqrt{5 4\cos 0} = r$
- $ho_{-}c$ decreases non-linearly from $r\sqrt{3}$ to r

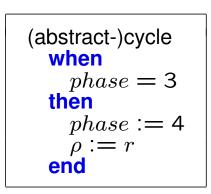
$$ho_{\scriptscriptstyle -} c(t) \;=\; r \sqrt{5-4\cos(rac{\pi}{3}-rac{v(t-now)}{r})}$$

Thus

$$rac{d
ho_{-}c(t)}{dt} = rac{4r\sin(rac{\pi}{3}-rac{v(t-now)}{r})}{2\sqrt{5-4\cos(rac{\pi}{3}-rac{v(t-now)}{r})}}rac{-v}{r}$$

When t increases from now to $now + \frac{\pi r}{3v}$, then the derivative $\frac{d\rho_{-}c(t)}{dt}$ increases monotonically from -v to 0:

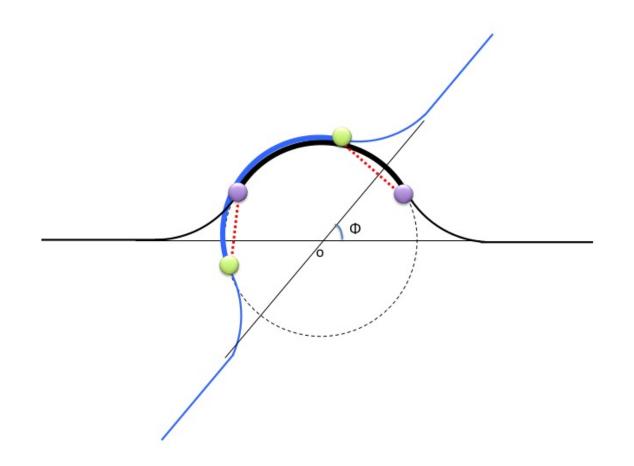
$$egin{array}{ll} rac{d
ho_{-}c(t)}{dt}_{t=now}&=&-oldsymbol{v}\ rac{d
ho_{-}c(t)}{dt}_{t=now+rac{\pi r}{3v}}&=&oldsymbol{0} \end{array}$$



(concrete-)cycle
when

$$phase = 3$$

then
 $phase := 4$
 $\rho_{-c} := \lambda t \cdot t \in now \dots now + \frac{2\pi r}{3v} | r$
 $now := now + \frac{2\pi r}{3v}$
end

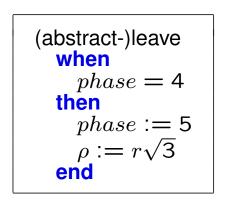


cycle when phase = 3then phase := 4 $\rho_{-}c := \lambda t \cdot t \in now \dots now + \frac{2\pi r}{3v} | r$ $now := now + \frac{2\pi r}{3v}$ end

 $ho_{-}c(now) = r$

$$-
ho_{-}c(now+rac{2\pi r}{3v}) = r$$

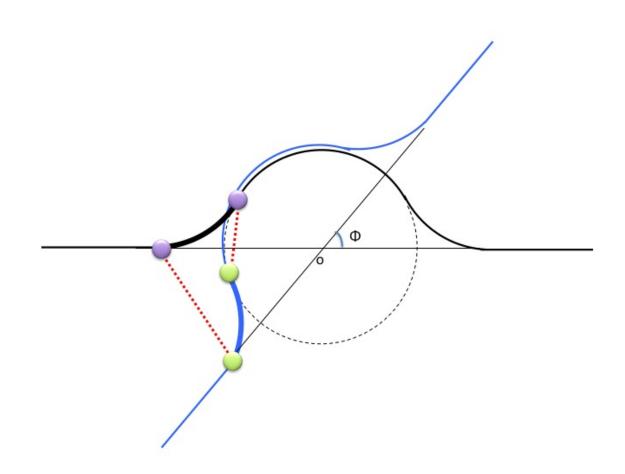
- $ho_{-}c$ remains constant to r



concrete-)leave
when

$$phase = 4$$

then
 $phase := 5$
 $\rho_{-c} := \lambda t \cdot t \in now ... now + \frac{\pi r}{3v} | r\sqrt{5 - 4\cos(\frac{v(t-now)}{r})}$
 $now := now + \frac{\pi r}{3v}$
end



leave
when

$$phase = 4$$

then
 $phase := 5$
 $\rho_{-}c := \lambda t \cdot t \in now \dots now + \frac{\pi r}{3v} | r\sqrt{5 - 4\cos(\frac{v(t-now)}{r})}$
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ho_-c(now) = r\sqrt{5-4\cos 0} = r$$

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- $ho_{-}c$ increases non-linearly from r to $r\sqrt{3}$

Derivation of $\rho_{-}c(t)$

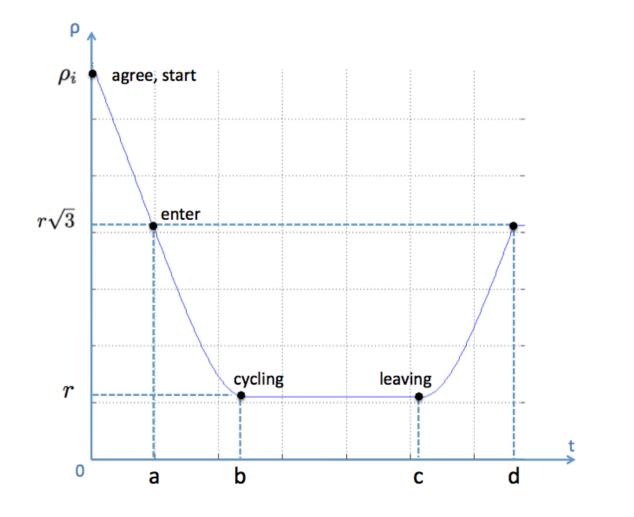
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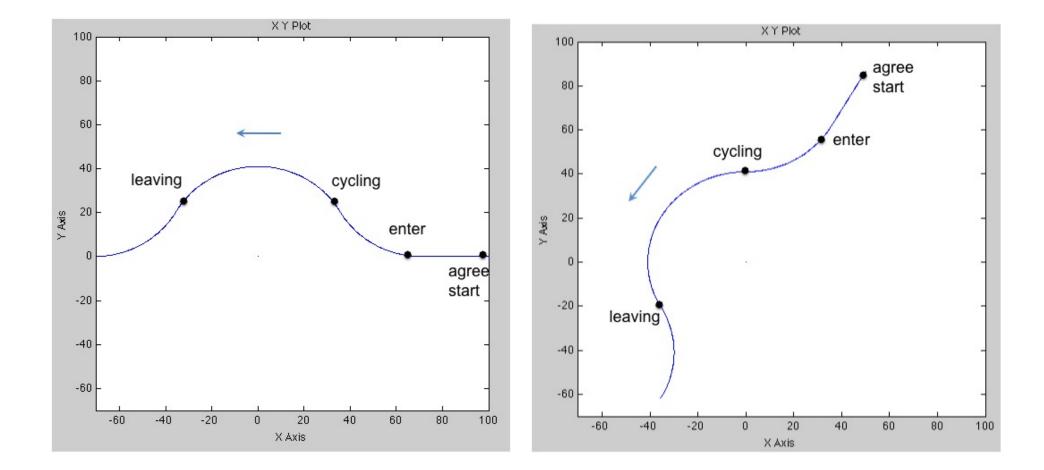
When *t* increases from *now* to $now + \frac{\pi r}{3v}$, then the derivative $\frac{d\rho_{-}c(t)}{dt}$ increases monotonically from 0 to v:

$$egin{array}{ll} rac{d
ho_{-}c(t)}{dt}_{t=now}&=&0\ rac{d
ho_{-}c(t)}{dt}_{t=now+rac{\pi r}{3w}}&=&oldsymbol{v} \end{array}$$

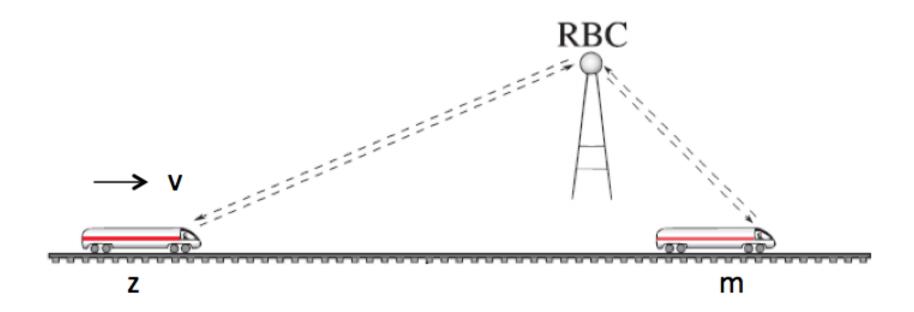


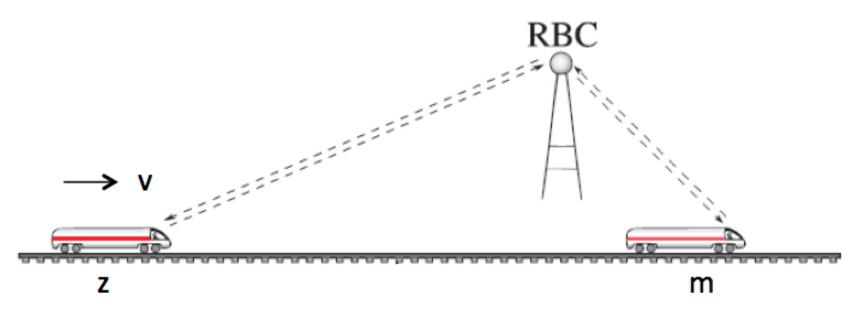
$$a=rac{
ho_i-r\sqrt{3}}{v}, \qquad b=a+rac{\pi r}{3v}, \qquad c=b+rac{2\pi r}{3v}, \qquad d=c+rac{\pi r}{3v}$$

Mathlab/Simulink Output 2: Aircraft Trajectories

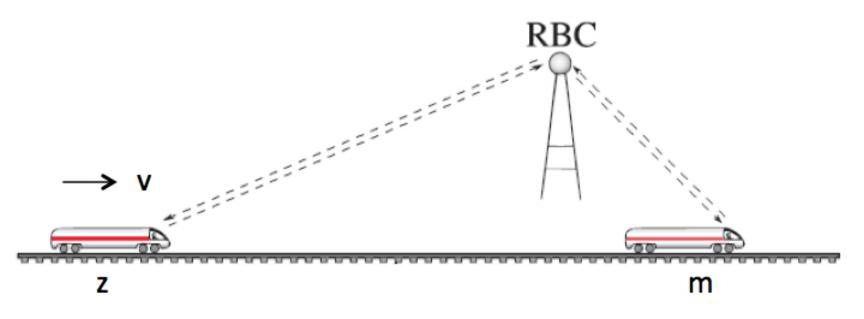


Example 2





- The second train is in position \boldsymbol{z}



- The second train is in position z
- It is made aware of a position m where it should at the latest stop

- The controller in the second train reacts every other ϵ seconds

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- Goal: Calculate the best acceleration at each controller's reaction.

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- This is the main invariant to be maintained

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$$2b(m-z-v\epsilon-arac{\epsilon^2}{2})\geq (v+a\epsilon)^2$$

- At each control time (every other ϵ seconds), the invariant to be maintained is:

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$$2b(m-z-v\epsilon-arac{\epsilon^2}{2})\geq (v+a\epsilon)^2$$

that is

$$2b(m-z) \geq v^2 + (a\epsilon^2 + 2v\epsilon)(a+b)$$

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- Otherwise, the acceleration should be -b (braking), resulting in:

$$2b(m-z) \geq v^2$$

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- We have the additional invariant: $v \in 0 \dots sl$
- We have thus three different controller decisions:
 - decision 1: acceleration -b
 - decision 2: acceleration A
 - decision 3: acceleration 0

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 - drive 1: if $v + a\epsilon \geq 0$ then $v + a\epsilon$
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- the new position of the train will be:
 - drive 1: if $v + a\epsilon \geq 0$ then $z + v\epsilon + arac{\epsilon^2}{2}$

- drive 2: if $v + a\epsilon < 0$ then $z + \frac{v^2}{2b}$ (the train stops after time $\frac{v}{b}$)

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- Continuous variables are not defined by differential equations

Thank you for listening