# Formalizing Hybrid Systems with Event-B 

Jean-Raymond Abrial, Wen Su, Huibiao Zhu

August 2012

## Prologue

## Using B Formal Method in Industry

## Train Applications

- Fully automatic train systems:
- Paris metro line 14 (October 1998)
- Roissy airport shuttle (March 2007)
- More train applications

| Line length | 8.5 km |
| :--- | :---: |
| Number of Stops | 8 |
| Time interval between two trains | 115 s |
| Speed | $40 \mathrm{~km} / \mathrm{h}$ |
| Number of trains | 17 |
| Passengers per day | 350,000 |


| Line length | 3.3 km |
| :--- | :---: |
| Number of Stops | 5 |
| Time interval between two trains | 105 s |
| Speed | $26 \mathrm{~km} / \mathrm{h}$ |
| Number of trains | 14 |
| Passengers per hour | 2,000 |


|  | Paris | Roissy |
| :---: | :---: | :---: |
| Number of final ADA lines (from B) | 86,000 | 158,000 |
| Number of proofs | 27,800 | 43,610 |
| Percentage of interactive proofs | 8.1 | 3.3 |
| Interactive proofs in Man.Month | 7.1 | 4.6 |

## Comparing the Case Studies (2)

- Man.month calculated with:
- 15 interactive proofs per man.day
- 21 days in a month
- In both cases, no unit tests and no integration tests
- Reinforcing global tests (catastrophic scenarios)
-Important differences in the software requirements:
- Paris: specially done for the project
- Paris: adaptation from O'Hare Airport (problems)

| City | Line | Service | Driverless |
| :---: | :---: | :---: | :---: |
| Algiers | 1 | 2011 | No |
| Barcelona | 9 | 2007 | Yes |
| Budapest | 4 | 2013 | Yes |
| Caracas | 4 | 2004 | No |
| Helsinki | 1 | 2013 | No |
| Hong Kong | TKO | 2001 | No |


| City | Line | Service | Driverless |
| :---: | :---: | :---: | :---: |
| Mexico | B | 2000 | No |
| New York | Canarsie | 2006 | No |
|  | PATH | 2014 | No |
| Paris | 14 | 1998 | Yes |
|  | 3 | 2009 | No |
|  | 1 | 2011 | Yes |
|  | 5 | 2012 | No |

## Similar Applications by Siemens (3)

| City | Line | Service | Driverless |
| :---: | :---: | :---: | :---: |
| Rennes | B | 2018 | Yes |
| Roissy CDG | 1 | 2007 | Yes |
| 2 | 2007 | Yes |  |
| San Juan | 2 | 2004 | No |
| Sao Paulo | TKO | 2001 | Yes |

Contact: [Jean-Marc.Meynadier@siemens.com](mailto:Jean-Marc.Meynadier@siemens.com)

| System | City | Service | Size | Language | Driverless |
| :---: | :---: | :---: | :---: | :---: | :---: |
| KVB | French Trains | 1993 | 30000 | ADA | No |
| CDTC | Cairo | 1996 | 3000 | Modula2 | No |
| SACEM | Paris (RER B) | 1996 | 2500 | Modula2 | No |
| ACSES | AMTRACK (USA) | 2002 | 14500 | ADA | No |


| System | City | Service | Size | Language | Driverless |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Urbalis 200 | Shanghai <br> New Dehli <br> Seoul | 2003 | 30000 | ADA | No |
|  | Daegu <br> Incheoun <br> Madrid <br> Santiago <br> Cairo | 2013 |  |  |  |
|  | Bangalore |  |  |  |  |
|  |  |  |  |  |  |
|  |  |  |  |  |  |

## Similar Applications by Alstom (3)

| System | City | Service | Size | Language | Driverless |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Urbalis 400 | Shanghai | 2008 | 100000 | ADA | No |
|  | Beijing |  |  |  | Yes |
|  | Chenzen |  |  |  | No |
|  | Sao Paulo | 2013 |  | Yes |  |
|  | Mexico |  |  | No |  |
|  | Milano |  |  | No |  |
|  | Toronto |  |  | No |  |
|  | Wuhan |  |  |  | No |

Contact: [Luis-Fernando.Mejia@transport.alstom.com](mailto:Luis-Fernando.Mejia@transport.alstom.com)

# Formalizing Hybrid Systems with Event-B 

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- Event-B is said to handle discrete transition systems: is it enough?
- Event-B is said to handle discrete transition systems: is it enough?
- Continuous transition systems are important too
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- Continuous transition systems are important too: are not they?
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- Continuous transition systems are important too: are not they?
- How can time be handled in Event-B?
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- Continuous transition systems are important too: are not they?
- How can time be handled in Event-B?
- Is it necessary to add a special "time feature" within Event-B?
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- Continuous transition systems are important too: are not they?
- How can time be handled in Event-B?
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Hybrid Systems

## Hybrid Systems

- The idea is then to introduce (somehow) continuous transitions


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- BUT, when introducing such continuous transitions the discrete transitions are still needed
- Hence the notion of hybrid systems
where both discrete and continuous transitions can occur
Typical Hybrid Systems ..... 25

Typical Hybrid Systems

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- A piece of software, the controller, manages an environment
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- Controller works from time to time in a DISCRETE fashion
- Hybrid frameworks are frequent in embedded systems where:
- A piece of software, the controller, manages an environment
- Controller is linked to environment by sensors and actuators
- Controller works from time to time in a DISCRETE fashion
- While environment evolves in a CONTINUOUS way.


## DISCRETE



CONTINUOUS

## DISCRETE



- We want to develop models of such closed systems


## DISCRETE



- We want to develop models of such closed systems
- We have thus to cope with both discrete and continuous evolutions


## Example

- Continuous physical environment:
a train defined by its position, speed, and acceleration


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a train defined by its position, speed, and acceleration



## Example (cont'd)

- Discrete controller:
a driver changing from time to time the acceleration of the train


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## Example (cont'd)

- Discrete controller:
a driver changing from time to time the acceleration of the train

- Goal: to control the speed of the train (station or another train)
R.J. Back and R. Kurki-Suonio.

Distributed Cooperation with Action Systems
ACM Transaction on Programming Languages and Systems. 1988.
R.J. Back, L. Petre, and I. Porres.

Generalizing Action Systems to Hybrid Systems.
FTRTFT 2000. LNCS 1926 Springer Verlag, 2000.
R.J. Back, C. Cerschi Seceleanu, and J. Westerholm.

Symbolic Simulation of Hybrid Systems.
APSEC'02, 2002.

## Our Papers

Formalizing Hybrid Systems with Event-B
ABZ Conference. Pisa, June 2012

Complementary Methodologies for Developing Hybrid Systems with Event-B

Accepted at ICFEM 2012. Kyoto, November 2012

## The Approach with Event-B (inspired by Action System)

- Discrete variables together with continuous variables
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- Continuous variables are time functions as in Action System
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- Discrete systems as an abstraction of continuous ones
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- Continuous variables are time functions as in Action System
- We are interested in the immediate future of continuous variables
- Discrete systems as an abstraction of continuous ones
- We thus use refinement from a discrete to a continuous system


## Three Examples

- The 2 examples:
- Aircraft collision avoidance
- Train control (time permitting),
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- Description:
- The problem,
- The constraints and goal,
- The solution,
- The discrete and continuous transitions
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- Aircraft collision avoidance
- Train control (time permitting),
- Description:
- The problem,
- The constraints and goal,
- The solution,
- The discrete and continuous transitions
- Examples developed and fully proved with the Rodin Platform
- These examples show complete analytical solutions


## Example 1

Aircraft Collision Avoidance

## Aircraft Collision Avoidance

- Two aircrafts are flying at the same altitude and speed


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d=2 \rho \sin \frac{\phi}{2}
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$$

- Their distance must always be greater than or equal to a constant $p$
- Goal: we want to find a solution to avoid the collision

- The radius $r$ of this circle will be determined later
- Both aircrafts continue to fly at the same speed during the maneuver
















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- Angle $\phi$ between aircrafts does not change during the maneuver


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- Angle $\phi$ between aircrafts does not change during the maneuver
- Both aircrafts are still at the same distance $r h o$ of the point $o$
- The only parameter that counts then in order to maintain the distance $\boldsymbol{d} \geq \boldsymbol{p}$ :

$$
d=2 \rho \sin \frac{\phi}{2} \geq p
$$

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- is the common distance $\rho$ of both aircrafts to the collision point $o$
- The smallest distance is when they are on the circle (more later)
- We must have then: $\frac{p}{2 \sin \frac{\phi}{2}} \leq r$

Making the Maneuver more Precise


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- We have then:

$$
\frac{p}{2 \sin \frac{\phi}{2}} \leq r \leq \frac{\rho_{i}}{\sqrt{3}}
$$

## Making the Maneuver more Precise

- Here is again the possible interval for the radius $r$ of the circle:

$$
\frac{p}{2 \sin \frac{\phi}{2}} \leq r \leq \frac{\rho_{i}}{\sqrt{3}}
$$

- We must have then the following for the constants $\rho_{i}, \phi$, and $p$ :

$$
2 \rho_{i} \sin \frac{\phi}{2} \geq p \sqrt{3}
$$

- $\phi$ is the angle of the two trajectories
- $\rho_{i}$ is the initial distance of the two aircrafts to the collision point $o$
- $p$ is the minimal safety distance between the two aircrafts
axm1: $\quad \phi \in 0 \ldots \pi$
axm2: $\quad \rho_{i} \in \mathbb{R}^{+}$
axm2: $\quad p \in \mathbb{R}^{+}$
axm4: $\quad 2 \rho_{i} \sin \frac{\phi}{2} \geq p \sqrt{3}$
- In this initial model, we are still discrete
- phase corresponds to the various discrete events
- $\rho$ is the common distance of the aircrafts to the collision point $o$
$-r$ is the circle radius
inv1: phase $\in\{0,1,2,3,4,5\}$
inv2: $\quad \rho \in \mathbb{R}^{+}$
inv3: $r \in \mathbb{R}^{+}$
inv4: $\quad 2 \rho \sin \frac{\phi}{2} \geq p$
- inv4 is the safety invariant: the minimal authorized distance is $p$
- INIT: initialisation
- agree: choose the radius of the circle
- start: start the maneuver
- enter: entering the circle
- cycle: move on the circle
- leave: leaving the circle

| INIT |
| :--- |
| begin |
| $\rho:=\rho_{i}$ |
| $p h a s e$ |
| $r$ |
| $r: \in \mathbb{R}^{+}$ |
| end |



```
agree
    any c where
        phase = 0
        p\leq2c\operatorname{sin}\frac{\phi}{2}
        c\sqrt{}{3}\leq\mp@subsup{\rho}{i}{}
    then
        phase := 1
        r:=c
    end
```


## Choosing the radius $r$ of the circle

| start |
| :--- |
| when |
| phase $=1$ |
| then |
| phase $:=2$ |
| $\rho:=r \sqrt{3}$ |
| end |

$\rho$ goes from $\rho_{i}$ to $r \sqrt{3}$


```
enter
when
    phase = 2
then
    phase:= 3
\rho
end
```



$$
\begin{aligned}
& \text { cycle } \\
& \text { when } \\
& \text { phase }=3 \\
& \text { then } \\
& \text { phase }:=4 \\
& \rho:=r \\
& \text { end }
\end{aligned}
$$



```
leave
when
    phase = 4
then
    phase:=5
\rho}:=r\sqrt{}{3
end
```



- We introduce the intermediate continuous parts
- We replace $\rho$ by $\rho_{-} c$ (that is $\rho$ continuous)
- We introduce now, the present time

$$
\begin{array}{ll}
\text { inv1_1: } & \rho_{-c} \in \mathbb{R}^{+} \rightarrow \mathbb{R} \\
\text { inv1_2: } & \text { now } \operatorname{dom}\left(\rho_{c} c\right) \\
\text { inv1_3: } & \rho=\rho c(\text { now }) \\
\text { inv1_4: } & \forall t \cdot t \in \operatorname{dom}\left(\rho_{-} c\right) \Rightarrow 2 \rho_{-} c(t) \sin \frac{\phi}{2} \geq p
\end{array}
$$

- inv1 3 is the gluing invariant
- inv1_4 generalises the previous invariant: $2 \rho \sin \frac{\phi}{2} \geq p$
(abstract-)INIT begin

$$
\begin{aligned}
& \quad \begin{array}{l}
\rho:=\rho_{i} \\
\text { phase } \\
r
\end{array}:=0 \\
& \text { end }: \in \mathbb{R}^{+}
\end{aligned}
$$

(concrete-)INIT begin

$$
\rho_{-} c:=\left\{0 \mapsto \rho_{i}\right\}
$$

$$
\text { phase }:=0
$$

$$
\boldsymbol{r}: \in \mathbb{R}^{+}
$$

$$
\text { now }:=0
$$

end
agree
any $c$ where
phase $=0$ $p \leq 2 c \sin \frac{\phi}{2}$
$c \sqrt{3} \leq \rho_{i}$
then

$$
\text { phase }:=1
$$

$$
r:=c
$$

end
(abstract-)start when
phase $=1$ then
phase $:=2$
$\rho:=r \sqrt{3}$ end
(concrete-)start when
phase $=1$ then
phase $:=2$
$\rho_{-} c:=\lambda t \cdot t \in$ now $\left.. . n o w+\frac{\left(\rho_{i}-r \sqrt{3}\right)}{v} \right\rvert\, \rho_{i}-v(t-n o w)$
now $:=n o w+\frac{\left(\rho_{i}-r \sqrt{3}\right)}{v}$ end

start
when
phase $=1$
then

$$
\text { phase }:=2
$$

$$
\rho_{c} c: \left.=\lambda t \cdot t \in n o w . . n o w+\frac{\left(\rho_{i}-r \sqrt{3}\right)}{v} \right\rvert\, \rho_{i}-v(t-n o w)
$$

$$
n o w:=n o w+\frac{\left(\rho_{i}-r \sqrt{3}\right)}{v}
$$

end

- $\rho c($ now $)=\rho_{i}$
$-\rho_{c} c\left(n o w+\frac{\left(\rho_{i}-r \sqrt{3}\right)}{v}\right)=r \sqrt{3}$
- $\boldsymbol{\rho}_{\boldsymbol{c}} \boldsymbol{c}$ decreases linearly from $\rho_{i}$ to $r \sqrt{3}$
$-\frac{\left(\rho_{i}-r \sqrt{3}\right)}{v}$ is the time it takes to fly from $\rho_{i}$ to $r \sqrt{3}$


## Computing $\rho$ During First Part of Maneuver

$$
\begin{aligned}
\rho^{2} & =a^{2}+b^{2} \\
& =r^{2}(1-\cos \alpha)^{2}+r^{2}(\sqrt{3}-\sin \alpha)^{2} \\
& =r^{2}\left(5-4 \cos \left(\frac{\pi}{3}-\alpha\right)\right) \\
\rho & =r \sqrt{5-4 \cos \left(\frac{\pi}{3}-\alpha\right)}
\end{aligned}
$$

- $\rho$ decreases from $r \sqrt{3}$ to $r$ when $\alpha$ goes from 0 to $\frac{\pi}{3}$.


## Computing $\rho$ During First Part of Maneuver



- The angle $\alpha$ increases from 0 to $\frac{\pi}{3}$ during this phase
- The distance is $\frac{\pi r}{3}$
- The time to cover this distance is thus $\frac{\pi r}{3 v}$
- We have: $\alpha=\frac{v(t-\text { now })}{r}$

```
(abstract-)enter
    when
        phase = 2
```

(concrete-)enter
when
phase $=2$
then
phase $:=3$
$\rho_{-} c:=\lambda t \cdot t \in$ now..now $+\frac{\pi r}{3 v} \left\lvert\, r \sqrt{5-4 \cos \left(\frac{\pi}{3}-\frac{v(t-n o w)}{r}\right)}\right.$
now $:=n o w+\frac{\pi r}{3 v}$
end


## enter

when

$$
\text { phase }=2
$$

then
phase := 3
$\rho_{-c}:=\lambda t \cdot t \in$ now $. . n o w+\frac{\pi r}{3 v} \left\lvert\, r \sqrt{5-4 \cos \left(\frac{\pi}{3}-\frac{v(t-n o w)}{r}\right)}\right.$
$n o w:=n o w+\frac{\pi r}{3 v}$
end

- $\rho c($ now $)=r \sqrt{5-4 \cos \frac{\pi}{3}}=r \sqrt{5-\frac{4}{2}}=r \sqrt{3}$
$-\rho_{-} c\left(n o w+\frac{\pi r}{3 v}\right)=r \sqrt{5-4 \cos \left(\frac{\pi}{3}-\frac{v \pi r}{r 3 v}\right)}=r \sqrt{5-4 \cos 0}=r$
- $\rho_{c} \boldsymbol{c}$ decreases non-linearly from $r \sqrt{3}$ to $r$

$$
\rho_{c} c(t)=r \sqrt{5-4 \cos \left(\frac{\pi}{3}-\frac{v(t-n o w)}{r}\right)}
$$

Thus

$$
\frac{d \rho_{-} c(t)}{d t}=\frac{4 r \sin \left(\frac{\pi}{3}-\frac{v(t-n o w)}{r}\right)}{2 \sqrt{5-4 \cos \left(\frac{\pi}{3}-\frac{v(t-n o w)}{r}\right)}} \frac{-v}{r}
$$

When $t$ increases from now to now $+\frac{\pi r}{3 v}$, then the derivative $\frac{d \rho c(t)}{d t}$ increases monotonically from $-v$ to 0 :

$$
\begin{aligned}
& \frac{d \rho}{d t}(t)_{t=n o w}=\quad-v \\
& {\frac{d \rho_{-} c(t)}{d t}}_{t=n o w+\frac{\pi r}{3 v}}=0
\end{aligned}
$$

(abstract-)cycle
(concrete-)cycle
when
phase $=3$
then
phase $:=4$
$\rho_{-} c:=\lambda t \cdot t \in$ now ..now $\left.+\frac{2 \pi r}{3 v} \right\rvert\, r$ now $:=n o w+\frac{2 \pi r}{3 v}$ end


## cycle

when

$$
\text { phase }=3
$$

then

$$
\text { phase }:=4
$$

$$
\rho c:=\lambda t \cdot t \in \text { now } \ldots \text { now } \left.+\frac{2 \pi r}{3 v} \right\rvert\, r
$$

$$
\text { now }:=n o w+\frac{2 \pi r}{3 v}
$$

end

- $\rho_{-c} \boldsymbol{c}($ now $)=r$
- $\rho_{-} c\left(n o w+\frac{2 \pi r}{3 v}\right)=r$
- $\boldsymbol{\rho} \boldsymbol{c} \boldsymbol{c}$ remains constant to $r$
(abstract-)leave when phase $=4$ then
phase $:=5$ $\rho:=r \sqrt{3}$ end

```
(concrete-)leave
    when
        phase \(=4\)
    then
        phase \(:=5\)
        \(\rho_{-} c:=\lambda t \cdot t \in\) now \(. . n o w+\frac{\pi r}{3 v} \left\lvert\, r \sqrt{5-4 \cos \left(\frac{v(t-n o w)}{r}\right)}\right.\)
        now \(:=n o w+\frac{\pi r}{3 v}\)
    end
```



## leave

## when

phase $=4$
then
phase $:=5$
$\rho_{\mathrm{C}} c:=\lambda t \cdot t \in$ now.. now $+\frac{\pi r}{3 v} \left\lvert\, r \sqrt{5-4 \cos \left(\frac{v(t-n o w)}{r}\right)}\right.$
now $:=$ now $+\frac{\pi r}{3 v}$
end
$-\rho c(n o w)=r \sqrt{5-4 \cos 0}=r$

- $\rho_{-} c\left(n o w+\frac{\pi r}{3 v}\right)=r \sqrt{5-4 \cos \frac{\pi}{3}}=r \sqrt{5-\frac{4}{2}}=r \sqrt{3}$
- $\rho \boldsymbol{c}$ increases non-linearly from $r$ to $r \sqrt{3}$

$$
\rho_{c} c(t)=r \sqrt{5-4 \cos \left(\frac{v(t-n o w)}{r}\right)}
$$

Thus

$$
\frac{d \rho_{-} c(t)}{d t}=\frac{4 r \sin \left(\frac{v(t-n o w)}{r}\right)}{2 \sqrt{5-4 \cos \left(\frac{v(t-n o w)}{r}\right)}} \frac{v}{r}
$$

When $t$ increases from now to now $+\frac{\pi r}{3 v}$, then the derivative $\frac{d \rho c(t)}{d t}$ increases monotonically from 0 to v :

$$
\begin{aligned}
& \frac{d \rho_{-c}(t)}{d t} t=n o w=0 \\
& {\frac{d \rho_{-} c(t)}{d t}}_{t=n o w+\frac{\pi r}{3 v}=v}=0
\end{aligned}
$$


$a=\frac{\rho_{i}-r \sqrt{3}}{v}, \quad b=a+\frac{\pi r}{3 v}, \quad c=b+\frac{2 \pi r}{3 v}, \quad d=c+\frac{\pi r}{3 v}$

## Mathlab/Simulink Output 2: Aircraft Trajectories




## Example 2

Train Control (adapted from A. Platzer's book)

## Train Control (adapted from A. Platzer's book)

- Two trains are sent some information by Radio BroadCasting
- Two trains are sent some information by Radio BroadCasting

- Two trains are sent some information by Radio BroadCasting

- The second train is in position $z$
- Two trains are sent some information by Radio BroadCasting

- The second train is in position $z$
- It is made aware of a position $m$ where it should at the latest stop


## Train Control: Constraints and Goal

- The controller in the second train reacts every other $\epsilon$ seconds
- The controller in the second train reacts every other $\epsilon$ seconds
- It can change the acceleration of the train according to 3 values:

Accelerations are: $\boldsymbol{A},-\boldsymbol{b}$, or 0 , where $\boldsymbol{A}$ and $\boldsymbol{b}$ are positive

- The controller in the second train reacts every other $\epsilon$ seconds
- It can change the acceleration of the train according to 3 values:

Accelerations are: $\boldsymbol{A},-\boldsymbol{b}$, or 0 , where $\boldsymbol{A}$ and $\boldsymbol{b}$ are positive

- The speed should never be greater than $s l$ (speed limit)
- The controller in the second train reacts every other $\epsilon$ seconds
- It can change the acceleration of the train according to 3 values:

Accelerations are: $\boldsymbol{A},-\boldsymbol{b}$, or 0 , where $\boldsymbol{A}$ and $\boldsymbol{b}$ are positive

- The speed should never be greater than $s l$ (speed limit)
- The train should never go backwards
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- The speed should never be greater than sl (speed limit)
- The train should never go backwards
- Goal: Calculate the best acceleration at each controller's reaction.

Train Control: Formal Reasoning

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- This is the main invariant to be maintained


## Train Control: Formal Reasoning (cont’d)

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- Otherwise, the acceleration should be -b (braking), resulting in:

$$
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$$

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- After the choice of acceleration, A or -b, the speed of the train is:

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- If $v+a \epsilon>s l$, we must choose a 0 acceleration (instead of A)
- We have the additional invariant: $v \in 0$.. sl
- We have thus three different controller decisions:
- decision 1: acceleration -b
- decision 2: acceleration A
- decision 3: acceleration 0


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- drive 2: if $\boldsymbol{v}+\boldsymbol{a} \boldsymbol{\epsilon}<\mathbf{0}$ then 0
- the new position of the train will be:
- drive 1: if $v+a \epsilon \geq 0$ then $z+v \epsilon+a \frac{\epsilon^{2}}{2}$
- drive 2: if $\boldsymbol{v}+\boldsymbol{a} \boldsymbol{\epsilon}<0$ then $z+\frac{v^{2}}{2 b}$ (the train stops after time $\frac{\boldsymbol{v}}{\boldsymbol{b}}$ )


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- The only thing that will be necessary in Event-B are Real Numbers
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- Continuous variables are not defined by differential equations


## Thank you for listening

